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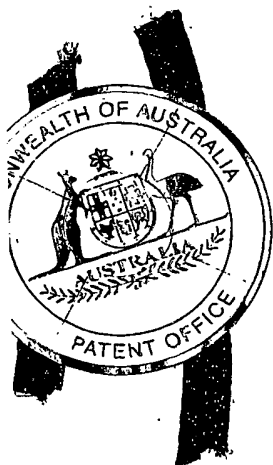
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I, JULIE BILLINGSLEY, TEAM LEADER EXAMINATION SUPPORT AND SALES hereby certify that annexed is a true copy of the Provisional specification in connection with Application No. 2002950004 for a patent by METAL STORM LIMITED as filed on 05 July 2002.

WITNESS my hand this
Sixteenth day of July 2003

A handwritten signature in cursive script, appearing to read "J. Billingsley".

JULIE BILLINGSLEY
TEAM LEADER EXAMINATION
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IGNITION ARRANGEMENT FOR STACKED PROJECTILES

Field of the Invention

5 The present invention relates to ignition arrangements for munitions and firearms. In particular, although not exclusively, the invention relates to fusing arrangements for igniting propellant charges associated with respective projectiles stacked axially within a barrel. The invention also relates to a method of igniting such propellant charges for stacked projectiles.

10 Background of the Invention

In earlier International Patent Applications No. PCT/AU94/00124 filed 14 March 1994 and No. PCT/AU96/00459 filed on 19 July 1996, assigned to the present applicant, there are disclosed several proposals for igniting propellant charges associated with projectiles stacked axially within a barrel. Whilst electronic ignition
15 arrangements for individual projectiles discussed therein provide for maximum flexibility for firing each projectile, some applications of this technology call for less complex ignition arrangements.

In particular, the applicant has now devised an ignition arrangement whereby a proposal for controlled burning of fuse columns extending through the projectiles
20 has been further advanced. We have now found a simplified ignition arrangement wherein a number of projectile in a barrel may be discharged at pre-determined intervals whilst obviating the requirement for individual electronic or mechanical ignition of each propellant charge in a barrel.

25 Disclosure of the Invention

According to a first aspect of the present invention there is provided an ignition arrangement for a barrel assembly including a barrel having a plurality of projectiles axially stacked within the barrel together with respective propellant charges for propelling the projectiles sequentially from the barrel, said ignition arrangement
30 including:

a fuse disposed in a cavity provided in a body of each projectile, wherein the cavity communicates both forwardly and rearwardly of the projectile body;

whereby, in use, said fuse burns at a controlled rate in the cavity and causes ignition of the propellant charge associated with said projectile.

The arrangement of the cavity is such that burning of said fuse is triggered by combustion of a forward propellant charge associated with the immediately preceding projectile in the stack as said preceding projectile is discharged.

Preferably the fuse is housed in a sleeve, suitably composed of metal. The fuse may comprise three sections, a forward section, a rear section and an intermediate section. The intermediate section suitably contains the material for controlled burning and is generally longer than said forward and rear sections.

The forward section of the fuse is composed of a fuse material ignitable by burning propellant via the first aperture, which ignited forward section can in turn ignite the intermediate section. In the case of the leading or forwardmost projectile in the barrel, the forward section of the fuse may alternatively be ignited by a primer using electrical or mechanical means, suitably via the first aperture.

The intermediate section is composed of material selected to burn at a pre-determined velocity and, at conclusion of the intermediate material burn, will be effective to ignite the rear section of the fuse. The fuse material of the intermediate section may also provide a sealing function, in that the burnt material provides at least a partial seal within the cavity and/or the first aperture.

The rear section is, accordingly, composed of a material that is ignitable by the burning intermediate section and is able, in turn, to ignite the associated propellant charge via the second aperture.

The pre-determined burn velocity is calculated with reference to the desired period of time between ignition of propellant charges. It will be appreciated that this calculation may need to account for the cross ignition times involving the forward section and the rear section of the fuse.

Preferably the projectile is provided with a first aperture that communicates between the cavity and forwardly of said projectile body and a second aperture that communicates between the cavity and rearwardly of the projectile body. Most preferably the cavity is aligned with an axis of the projectile, and suitably extends coaxially of said projectile. The first and second apertures are generally restricted in size compared with the cavity, the first aperture preferably being smaller than the second aperture.

In a further aspect of the invention, there is provided a projectile including a body portion having a cavity containing a fuse and wherein the projectile is provided with a first aperture that communicates between the cavity and forwardly of said projectile body and a second aperture that communicates between the cavity and rearwardly of the projectile body.

The projectile body is desirably composed of at least two separate components to facilitate convenient insertion of the fuse into the cavity, which components are suitably fixed together subsequent to fuse insertion thereby retaining the fuse within the projectile body. If required, the projectile components may be releasably coupled together.

The two components of the projectile body may include a head member suitably composed of steel and a tail member, suitably composed of aluminium and including a trailing skirt portion.

The propellant charge is suitably formed as a block, and may be contained within the trailing skirt portion.

The skirt portion may, in use, be engaged by a mandrel for urging an outer face of the trailing skirt portion into operative sealing engagement with the bore of the barrel. The mandrel may be formed by the propellant charge or by the head member of a trailing projectile. In another form, the nose portion of a trailing projectile may be urged into operative sealing engagement with an inner end face of the trailing skirt portion.

Alternative projectile configurations suited to provision of a fuse cavity and envisaged in earlier patent applications by the present applicant, including those described in International Application No. PCT/AU98/00409 and PCT/AU98/00414, are hereby expressly incorporated by reference.

In another aspect, the present invention provides a method of igniting a plurality of propellant charges associated with respective projectiles axially stacked with a barrel, wherein a fuse is disposed in a cavity provided in a body of each projectile, wherein the projectile is provided with a first aperture that communicates between the cavity and forwardly of said projectile body and a second aperture that communicates between the cavity and rearwardly of the projectile body, said method including the steps of:

igniting the fuse in the leading or forwardmost projectile in the barrel whereby, said fuse burns at a controlled rate in the cavity and causes ignition of the propellant charge associated with said projectile;

which ignited propellant charge propels the leading projectile from the barrel
5 and ignites the fuse contained in the next projectile in said stack;
whereby the remaining projectiles of said plurality of projectiles are subsequently propelled from the barrel in sequence.

In yet another form of the invention, there is provided a weapon including a
10 cluster of barrel assemblies each having an ignition arrangement as set out above.

Brief Details of the Drawings

In order that this invention may be more readily understood and put into practical effect, reference will now be made to the accompanying drawings which
15 illustrate a typical embodiment of the invention and wherein:-

FIG. 1 is a schematic cross-sectional view of a barrel assembly incorporating an ignition arrangement of a first embodiment; and

FIG. 2 is a further schematic view of a projectile incorporating a fuse of the first embodiment.

20

Description of Preferred Embodiments

Referring to FIG. 1 there is shown a barrel assembly 10 including a barrel 11 having a muzzle end 12, having a barrel end cover 13 in place over the muzzle, and a rear end 14. The end cover 13 is constructed of plastics or cellulose material and
25 prevents ingress of foreign matter in the barrel 11. In some applications of the invention, such as perimeter defence, a weapon employing such barrel assemblies will be exposed to weather for extended periods.

The barrel contains a plurality of projectiles 15a, 15b, 15c, each having a body 18 comprising a head member 16 and a tail member 17. The tail member 17 further
30 includes a rearwardly extending skirt portion 19 that abuts a rearward projectile in the present embodiment. Further details of the projectile 15, particularly in relation to a cavity 27 provided in the projectile body 18, are described below in relation to FIG. 2.

The projectiles 15 are axially stacked within the barrel together with respective propellant charges 20, which are arranged within the skirt portion 17 of each projectile. The propellant charges 20 are formed as a solid block, each with a graduated weight in the embodiment. The graduated weight of propellant charges
5 allows the muzzle velocity of the projectiles 15 in the stack to be varied. For example, in some applications, it may be desirable for the projectiles to arrive at a target almost simultaneously.

An inner end face 30 of the skirt portion 17 is engaged by a nose portion of a trailing projectile, which acts as a mandrel, urging an outer face 33 of the trailing skirt
10 portion 17 into operative sealing engagement with the bore of the barrel 11. In an alternative embodiment, the tail member of the projectile may be more rigid, and have a complementary face to that of the head of a trailing projectile, whereby sealing engagement may be formed between projectiles. This sealing arrangement is described further in our co-pending Provisional Application No. PS1828 dated 19
15 April 2002.

A cavity 27 is provided in each projectile body 18 for containing a fuse 22 that is part of the ignition arrangement for the propellant charges 20. The cavity 27 is cylindrically shaped and co-axially located in the projectile, and communicates both forwardly and rearwardly of the projectile body 18.

20 In the embodiment, the leading or forwardmost projectile 15a in the barrel contains a starter fuse 21 which is adapted for ignition by an electrical signal supplied via an ignition circuit 23. However the fuses 22 contained in the remaining projectiles 15b, 15c are adapted to be ignited by combustion of the propellant charge 20 associated with the projectile (ie. 15a, 15b) immediately forward of the projectile in
25 question. It should be noted that a single electrical signal is all that is required to initiate discharge of the plurality of projectiles 15 from the barrel 11.

Other embodiments may employ chemical or mechanical initiation of the fuses, and may also include separately initiated chains of projectiles in a single barrel. For example, one barrel containing 12 projectiles fused in three (3) groups of
30 4 projectiles may be served by three(3) circuits which initiate each group as desired. A variation on the ignition circuit is the use of a single circuit which delivers a coded ignition signal to all three (3) starter fuses, wherein each starter responds to a specific code.

Referring particularly to FIG. 2, the head 16 and tail 17 components of the projectile body 18 are arranged to be assembled together subsequent to insertion of the fuse 22 into the co-axial cavity 27. The head member 16 includes a socket portion 31 which is engaged by a spigot portion 32 of the tail member 17, which might be screw threaded to facilitate subsequent release, but in any event can be fixed together to retain the fuse 22 therein. In the embodiment, the head member 16 is composed of steel, whilst the tail member 17 is composed of an aluminium alloy.

When assembled, the co-axial cavity 27 communicates forwardly of the projectile body 18 via a front aperture 28 conveniently formed in the head member 16. The co-axial cavity 27 also communicates rearwardly of the projectile body 18 via a rear aperture 29 conveniently formed to the tail member 17. In particular the rear aperture exits the tail member in the vicinity of the propellant charge 20 associated with the projectile 15.

The front aperture 28 is quite restricted in size compared with the cavity 27 in order to minimise passage of combustion products and loss of pressure when a forward propellant charge combusts. The rear aperture is also relatively restricted, but not to the same extent as the front aperture in order to facilitate ignition of the rearward propellant charge 20 (shown in phantom). Both apertures 28, 29 are desirably smaller in size than the fuse 22 in order to retain the fuse within the cavity 27.

The fuse 22 is constructed of three sections in the embodiment, a front section 24, an intermediate section 25 and a rear section 26. The front section 24 is composed of a fuse material which is capable of being ignited by a forwardly disposed combustive propellant charge through the front aperture 28. In contrast, the rear section 26 of the fuse 22 is composed of a fuse material which is capable of igniting a rearwardly disposed fresh propellant charge through the second aperture 29.

More importantly, the intermediate section 25 of the fuse 22 is composed of a fuse material which is sized and formulated to burn at a predetermined velocity along its length, and also to ignite and be ignited by respective rear and front sections of the fuse. Suitable fuse material having these characteristics may be sourced from The Ensign Bickford Company of Simsbury, CT. The burn velocity of the fuse material in the intermediate section is chosen in light of the desired time between

ignition of propellant charges and consequent projectile discharges. It is envisaged that the fuse link construction of the invention will facilitate rates of fire of, for example 600 rounds per minute (rpm), 60,000 rpm and perhaps up to 600,000 rpm in a multiple barrel configuration.. The rate of fire from a single barrel is typically
5 expected to be between 300 and 45,000 rpm.

In another embodiment of the invention, the composition of the intermediate material is selected, such that when burnt, the residual melted material (perhaps a filler) provides a sealing action in the front aperture 28 and/or across the cavity 27 to further mitigate pressure loss. Alternatively, the fuse may be provided with a housing
10 which melts and obturates the cavity 27 during or consequent to fuse burning.

The front and rear sections capping the ends of the fuse 22 are provided because it is believed that the fuse material suited to controlled burning may be unable to reliably initiate the propellant charges 20 and vice-versa. Where a suitably synergistic fuse material and propellant can be specified, the front and rear sections
15 capping the fuse may not be required. In some embodiments, the sections may include an initiating additive in quantities graded longitudinally from each end of the fuse.

Certain embodiments of the ignition fuse arrangement of the invention provide a several advantages, including:

20 1. The very greatly reduced opportunity for a following projectile to be ignited without the leading projectile having been previously ignited. It is believed to be almost impossible for a projectile to be 'skipped' and for a following or rearward projectile to be ignited when projectile(s) are still in forward of the projectile in question that haven't yet been ignited.

25 2. The system also reduces the number of wire loops from one for each projectile in existing electrical initiation systems, wherein conductors are provided for each propellant charge, to one conductor for each barrel. This means that a barrel assembly utilising the invention is much less prone to electrical failure as well as being lighter and smaller.

30 3. Pursuant to 2 above, the barrel assembly requires a very much smaller and simpler fire control unit.

4. Finally, the ignition arrangement reduces cost of manufacture by a significant degree.

Although it will be appreciated that initiation of the leading propellant charge will result in the discharge of a plurality of projectiles from the barrel through the chain of fuses 22 and projectile charges 20, there are many situations where this need not be a disadvantage.

5 The centre-fuse system can be incorporated into barrel assemblies, whether of existing or proposed configuration. For example, a number of these barrel assemblies employing the ignition arrangement of the invention can quite easily be clustered together in a 'pod' configuration, such as described in the above referenced patent applications. Whilst an accepted benefit of having more than one barrel is that
10 a pod type weapon system becomes repeatable, and different barrels may be loaded with different types and numbers of projectiles.

Accordingly, any application wherein a fixed number of projectiles are to be fired in any one instance is suitable for the invention. A specific example of such an application is vehicle self-defence against shoulder launched anti-tank missiles. In
15 this case a pre-determined number of projectiles can be fired from barrels in a pod in order to achieve the requisite high degree of probability of interception of the missile. Such pods as these can be placed on a vehicle, such as a tank, to deal with the threat of shoulder launched anti-tank missiles.

Anti-tank missiles will typically be launched at a vehicle from relatively close
20 ranges, such as 0.5 Km for example. The difficulty for the tank is that the time of flight of the missile is therefore very short, and although electronic detection of the threat can be almost immediate, it is very difficult for a vehicle self defence system, such as a missile or machine gun, to react rapidly and effectively enough to deal with the treat.

25 In the case of a machine gun, by the time it turrets onto the threat and starts firing, it will be lucky to get a few rounds fired at best. However, a compact and lightweight pod can turret more rapidly. The pod can fire from numerous barrels which can optionally be splayed to open up the pattern of fire at ultra rapid rates to produce a 'cone' of projectiles toward the missile. This contrasts with machine gun
30 fire which can deliver a point-on-point impact of only one very slow line of fire. With barrel splay, the pod can engage a target fired at close range with numerous rounds even before the centre line of the pod has completed turreting onto the target.

It will of course be realised that the above has been given only by way of illustrative example of the invention and that all such modifications and variations thereto as would be apparent to persons skilled in the art are deemed to fall within the broad scope and ambit of the invention as is herein set forth.

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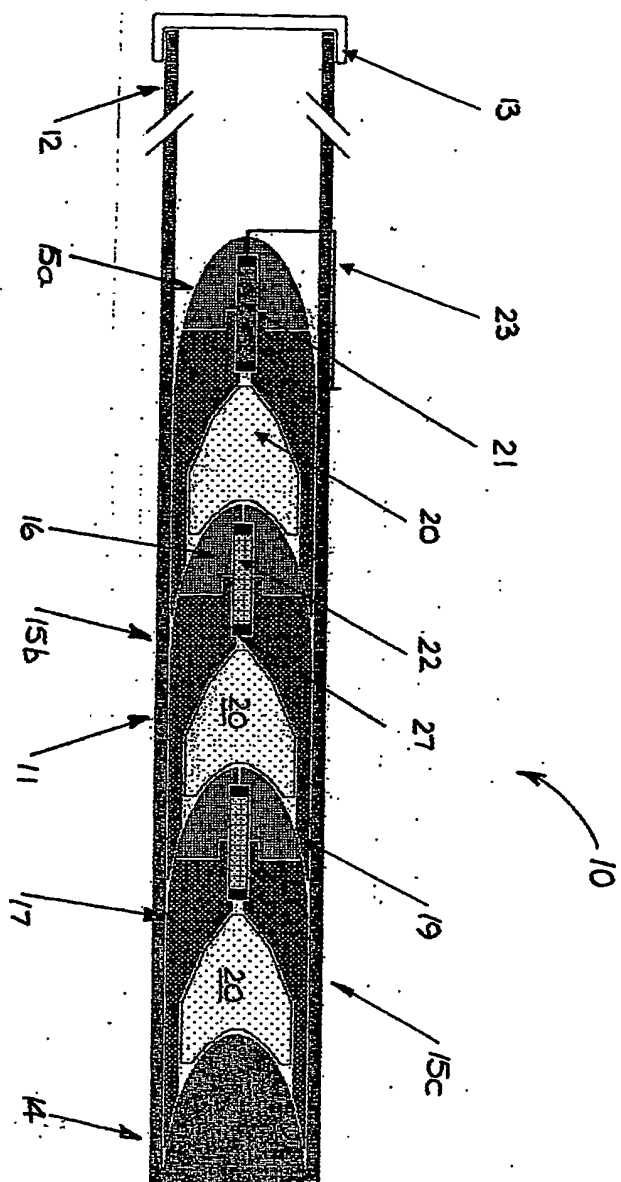
METAL STORM LIMITED

by

PIZZEYS Patent and Trade Mark Attorneys

ABSTRACT

An ignition arrangement for a barrel assembly (10) including a barrel (11) having a
5 plurality of projectiles (15a, 15b, 15c) axially stacked within the barrel (11 together
with respective propellant charges (20) for propelling the projectiles sequentially from
the barrel, said ignition arrangement including a fuse (21, 22) disposed in a cavity
(27) provided in a body (18) of each projectile, wherein the cavity (27) communicates
both forwardly (28) and rearwardly (29) of the projectile body (18); whereby in use,
10 said fuse (21) burns at a controlled rate in the cavity (27) and causes ignition of the
propellant charge (20) associated with said projectile (15a), which in turn ignites the
next following fuse (22) associated with a trailing projectile (15b).



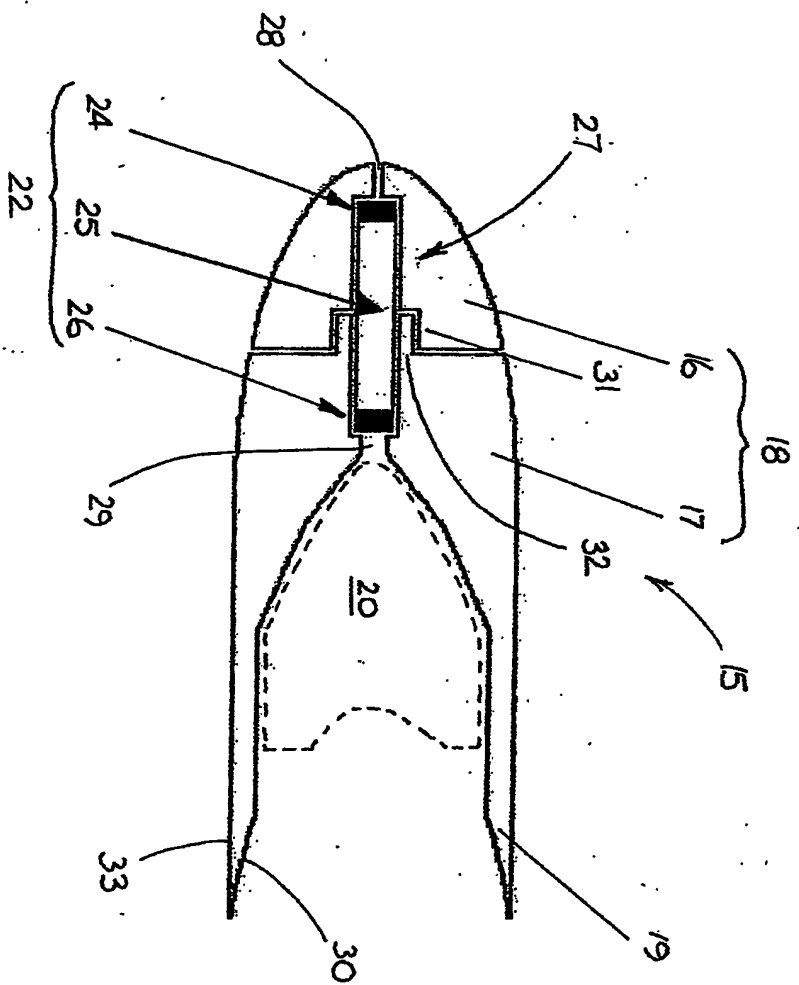


FIG. 2

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